



SKA Dish Array

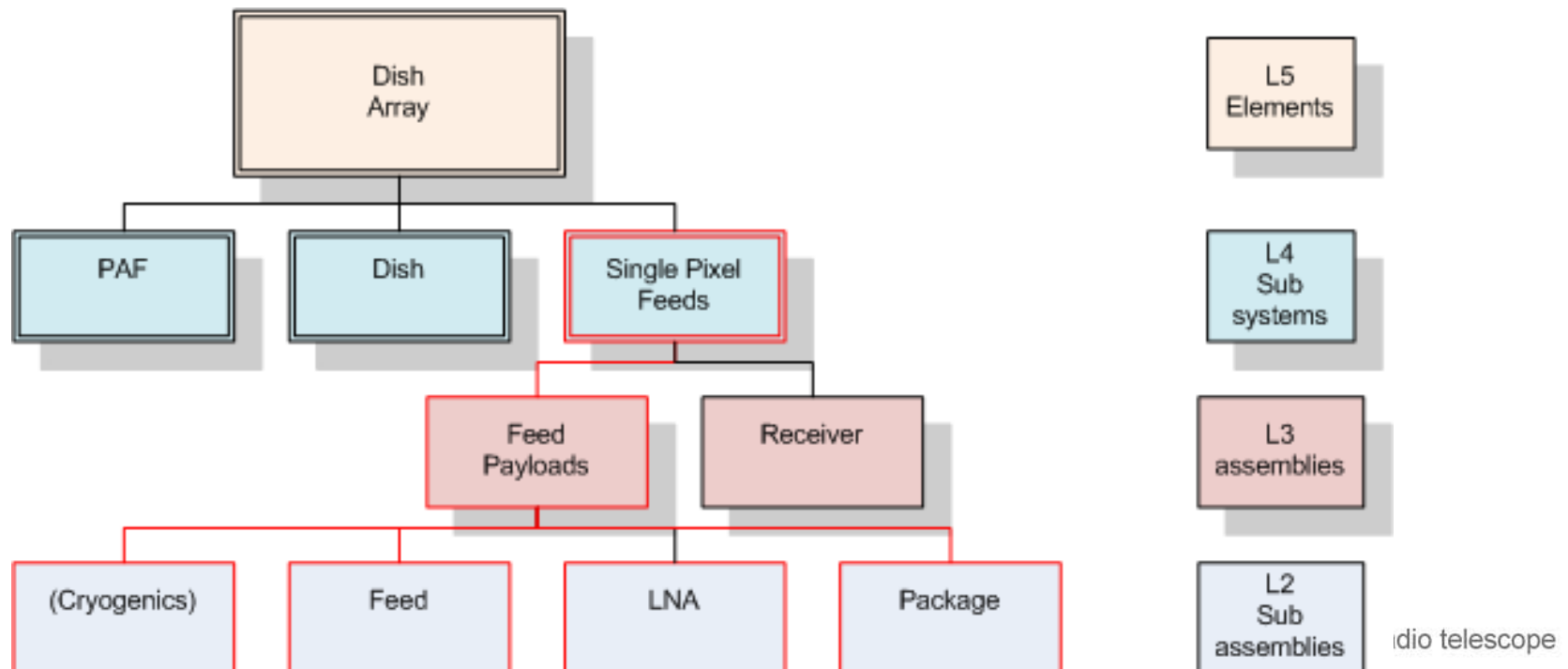
P. Dewdney

WP2 meeting
Oct 18, 2011

Dish Array Hierarchy



- SPF payloads comprise feed (including OMT if needed), LNAs, cryogenics (if needed) and packaging (dewar or other housing)



Key Requirements: Frequency range



- SKA1 requires 0.45 to 3 GHz
- SKA2 extends this to 10 GHz: this has implications for the dish design
 - Dish performance must meet specifications up to 10 GHz (SKA1 dishes will not be replaced in SKA2).
 - Dishes must be capable of accommodating feed payloads to cover 0.45 (possibly ~ 0.30 MHz) to 10 GHz.

Key Requirements: Sensitivity ($A_{\text{eff}}/T_{\text{sys}}$)



- The Phase 1 Dish Array shall have a sensitivity of $10^3 \text{ m}^2 \text{ K}^{-1}$ in the frequency range 450 MHz - 3 GHz.
- SKA2 requirement is $10^4 \text{ m}^2 \text{ K}^{-1}$ up to 10 GHz:
 - The aim will be to maximize sensitivity per €/ \$ *total system cost of ownership* whilst meeting other requirements.

Key Requirements:

Imaging dynamic range



- The SKA2 system requirement is for an **imaging dynamic range** for continuum of at least **74 dB** at **1.4 GHz**.
 - This requires the dish to have extremely stable, predictable beam shape and pointing in typical environmental conditions.
 - Stability and calibratability of the complete signal chain are also vital.
- Instrumental Pol'n:
 - -40 dB on axis after calibration
 - -30 dB to $\frac{1}{2}$ power point after cal (TBC).
- *These are a whole system requirements, but the dish array performance must not be limiting.*

Key Requirements: Dish Array operating cost



- There is no one specific requirement, but there are multiple contributors:
 - power consumption, repair period, continuous operation period
 - Thousands of dish systems will be very expensive to operate unless they are designed for high reliability with minimum maintenance.
 - Feeds must be individually maintainable/replaceable.
 - The maintenance regimes at existing radio astronomy observatories will not be affordable on this scale.
 - Routine maintenance intervals of at least one year are required, including dish mechanics and cryogenics.
 - Power consumption is a huge challenge for the SKA and the Dish Array is potentially a big contributor.

Key Requirements: Upgradeability and feed flexibility



- The Phase 1 Dish Array shall be upgradable.
 - Upgrades may include addition and replacement of single pixel payloads and receivers as well as the addition of phased array feeds.
- Multiple single pixel feeds and a phased array feed are to be accommodated.
 - A significant means of improving overall SKA system performance will be obtained through enhancement of feeds and receivers, especially in the transition of SKA1 to SKA2.

Key Requirements:

Minimum life time



- The Phase 1 Dish Array shall be designed for a **minimum life time of 30 years**, including initial installation, testing and commissioning period.
- Life-time extension
 - Large scale maintenance and/or an upgrade shall give the possibility to reach a life time of 50 years (TBC).

Some other important aspects of the Dish Arrays



- Some other aspects not yet explicitly covered in the requirements documents are as follows:
 - **Mass manufacture:**
 - Innovative manufacturing techniques will be needed to allow cost effective production of 15 m dishes in quantities of thousands.
 - **Rapid installation:**
 - Dish systems will need to be installed rapidly using minimal on-site manpower and equipment.
 - This is to minimize the impact on observations, as well as keep down the manpower cost.

Options under consideration: Dishes and Payloads



1. Dishes

- a) US TDP/DRAO
- b) NAOC/JLRAT/CETC54
- c) ASTRON/Airborne
- d) NRF – MeerKAT design

2. Feeds/Receivers and other “payloads”

- a) Single-pixel Feeds (corrugated horns)
- b) Wide-band SPFs (to be discussed under AIP session)
- c) Phased Array Feeds (PAFs) (to be discussed under AIP session)
- d) Receivers (various types of receiver systems)
- e) Other equipment: digitizers, optical modulators

Features of Dish Designs/Options



1. **US TDP/DRAO Dish**
 - a) Offset Gregorian optics
 - b) Offset high
 - c) Single piece main reflector, CFRP composite
 - d) Mold-based reflector fabrication, in-place
 - e) Evaluated with various SPF feeds and PAF.
2. **ASTRON/Airborne/Chalmers Dish**
 - a) Thermoplastic composite welded panels
 - b) Symmetric optics
 - c) Quasi-single piece main reflector, CFRP composite
 - d) Welded reflector fabrication on mold, in-place
 - e) Evaluated with Eleven feed (>1.2 GHz only).

Features of Dish Designs (cont'd)



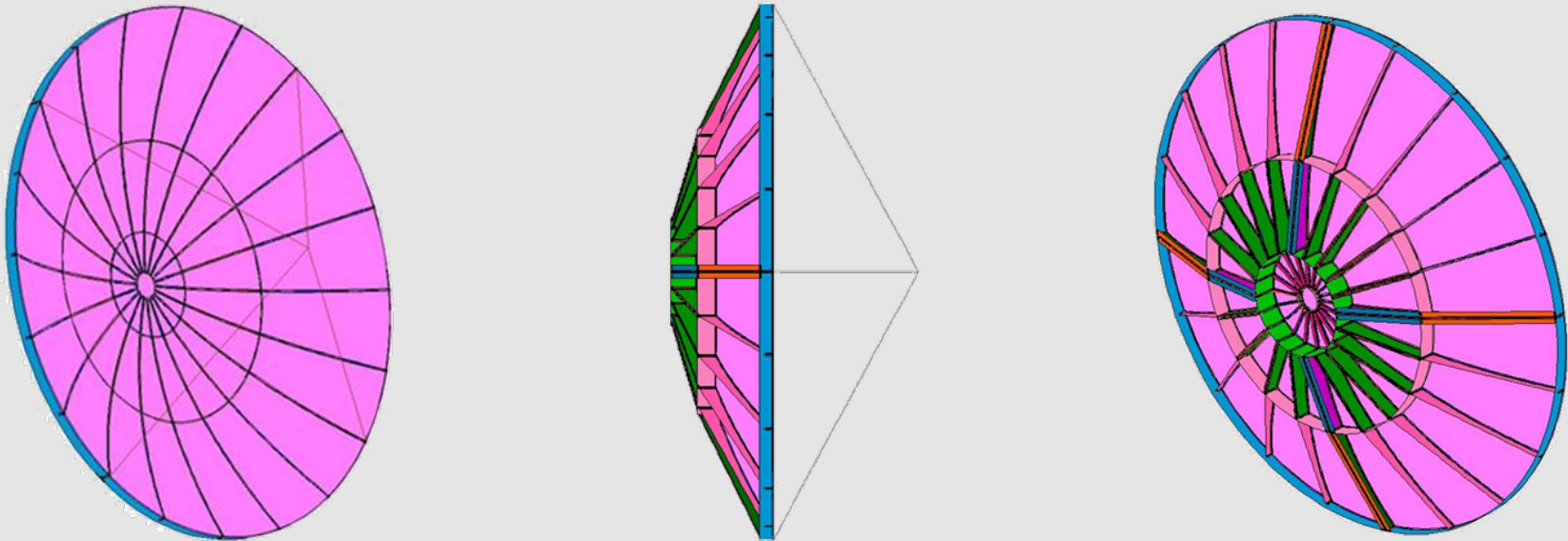
3. **NRF – MeerKAT Dish**
 - a) Offset Gregorian optics
 - b) Offset low
 - c) Single piece main reflector, Fibreglass composite
 - d) Mold-based reflector fabrication, in-place
 - e) Corrugated horn feed(s)
 - f) Smaller than required for SKA (13.5 m).
4. **NAOC/JLRAT/CETC54 (many combinations)**
 - a) Offset or Symmetric.
 - b) Alt-Az or 3-axis.
 - c) Composite or metal
 - d) Evaluated with similar to Eleven feed.

Thermoplastic Reinforced Composite Reflector



Structural Design

- Baseline design reflector:
 - a stiffened skin with several different stiffeners.
 - entire structure same thermoplastic carbon based material.
 - manufactured using a single automated production process.



Symmetric Dish Thermoplastic Reinforced Composite



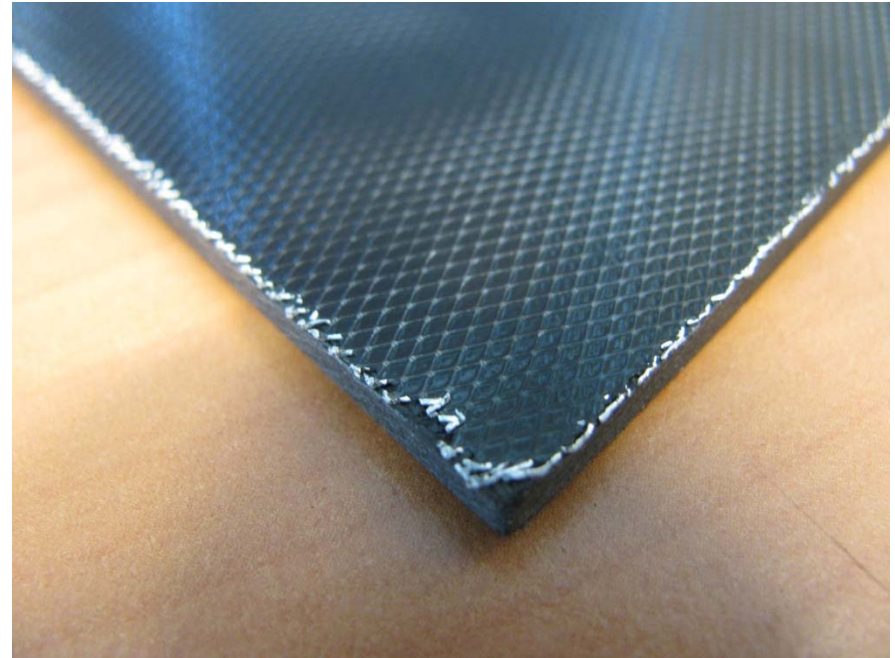
13/15 July '11

Thermoplastic carbon reinforced composite material

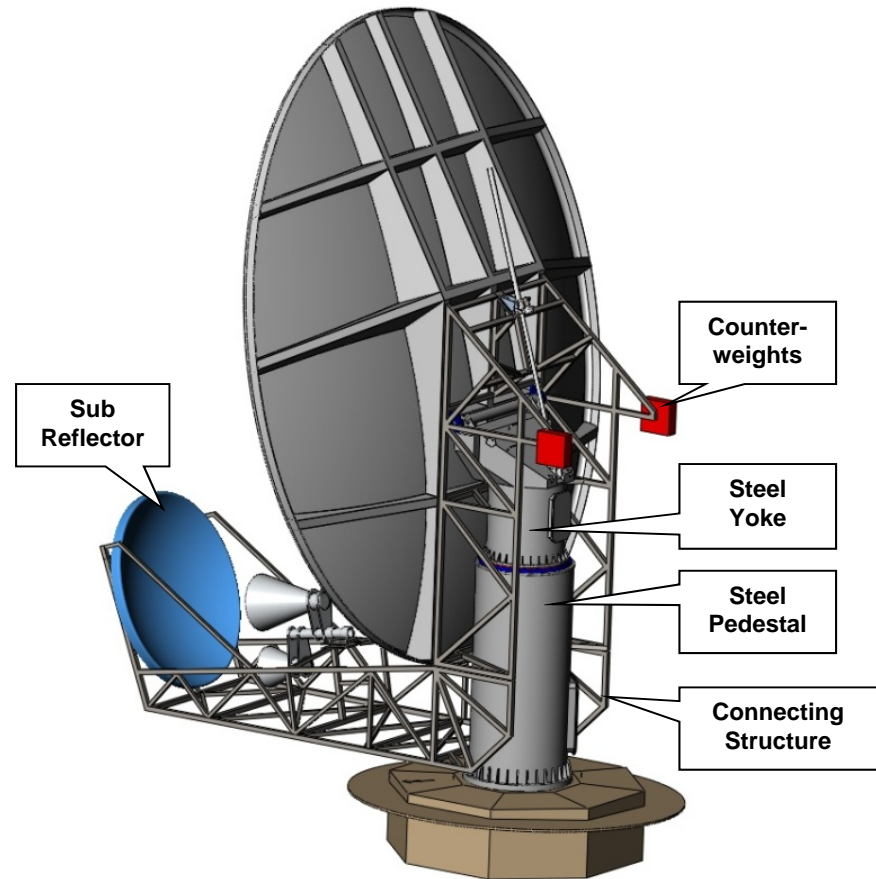
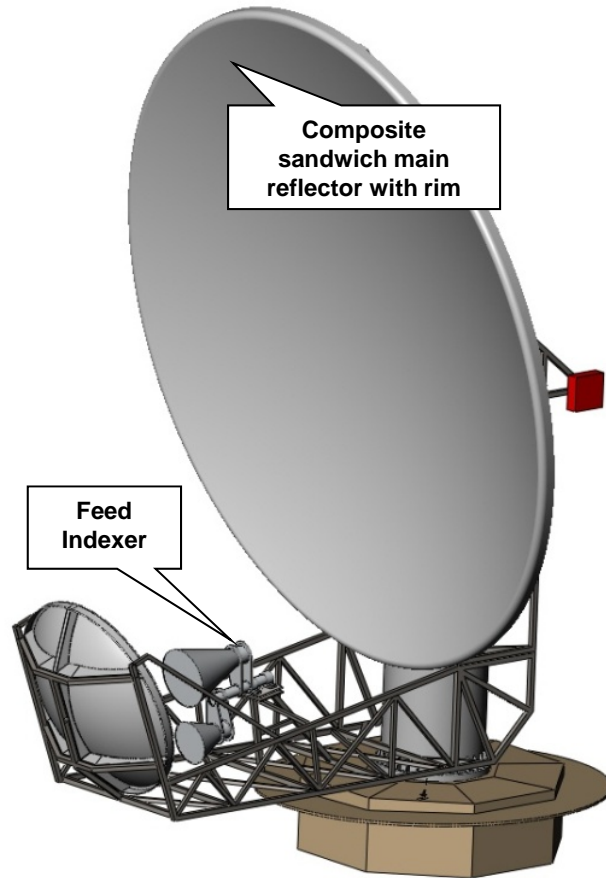
Benefits

- Embed a thin metal mesh to add reflectivity performance
- Low thermal coefficient
- Low weight (-30%: metal option)
- Coating to protect from atmospheric influences easily applied
- Suitable for recycling and repairs

- Thermoplastic composite
 - tougher, more ductile and robust compared to metal options,
 - Combined with carbon fibres outperforms aluminium and steel constructions
- **Reflectivity:** Embedded thin metal mesh for good reflectivity; initial studies show >99% reflectivity



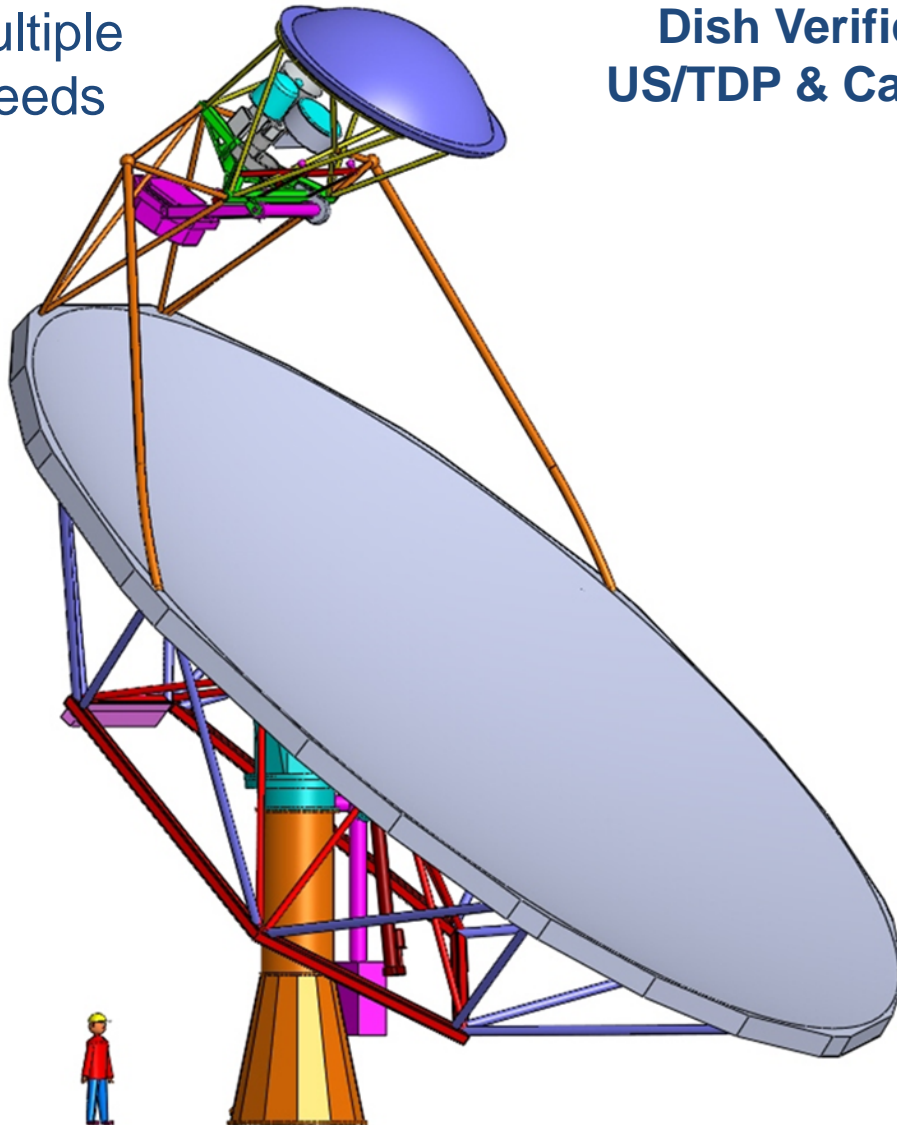
MeerKAT Dish Concept



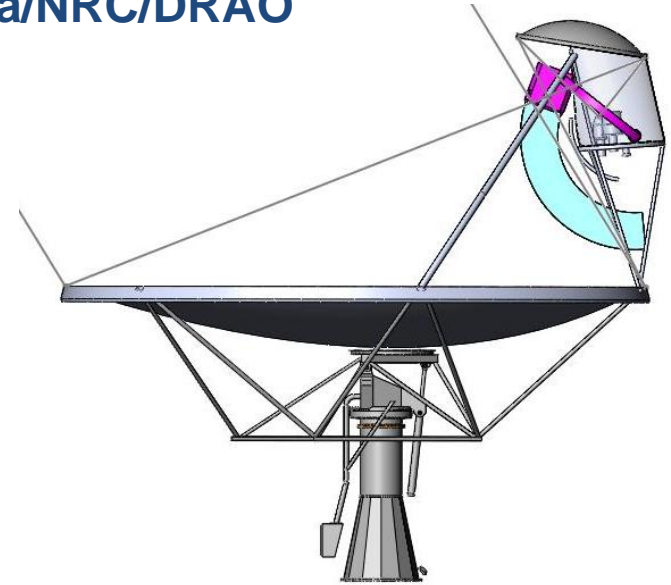
Offset-Optics Antenna Design



Multiple Feeds



Dish Verification Antenna
US/TDP & Canada/NRC/DRAO

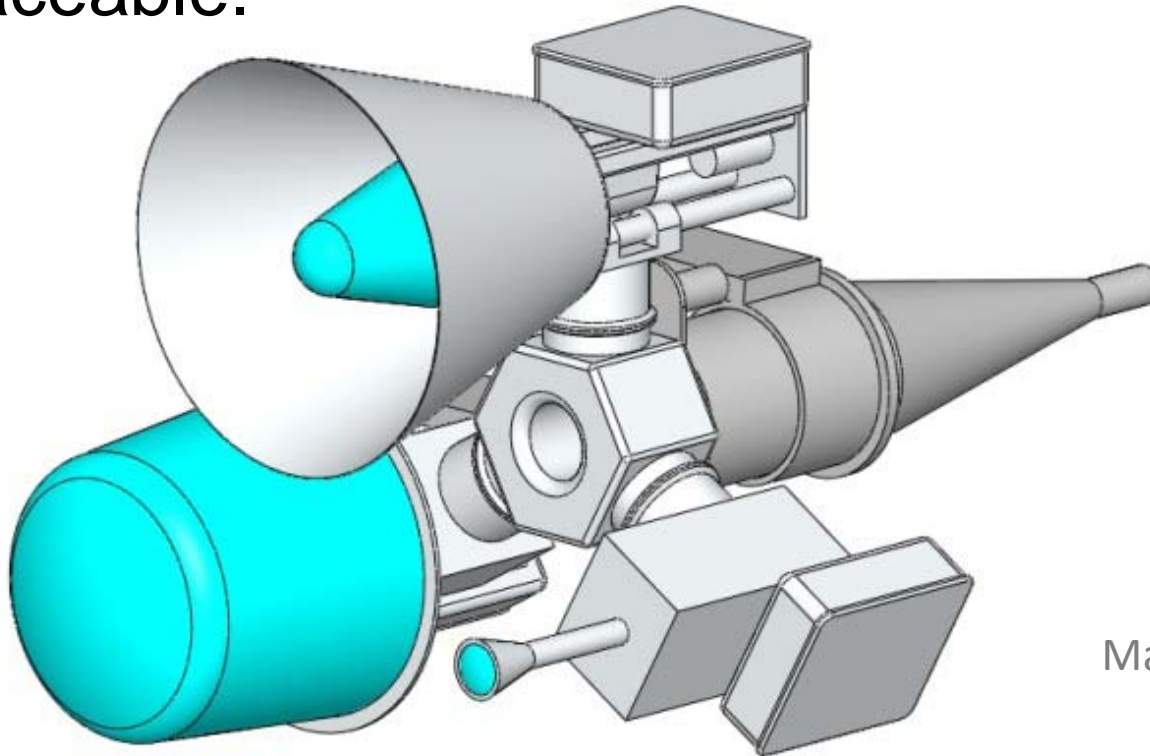


Deployment of PAFs.
Mid-freq. Wide-Field.

SPF feed payloads

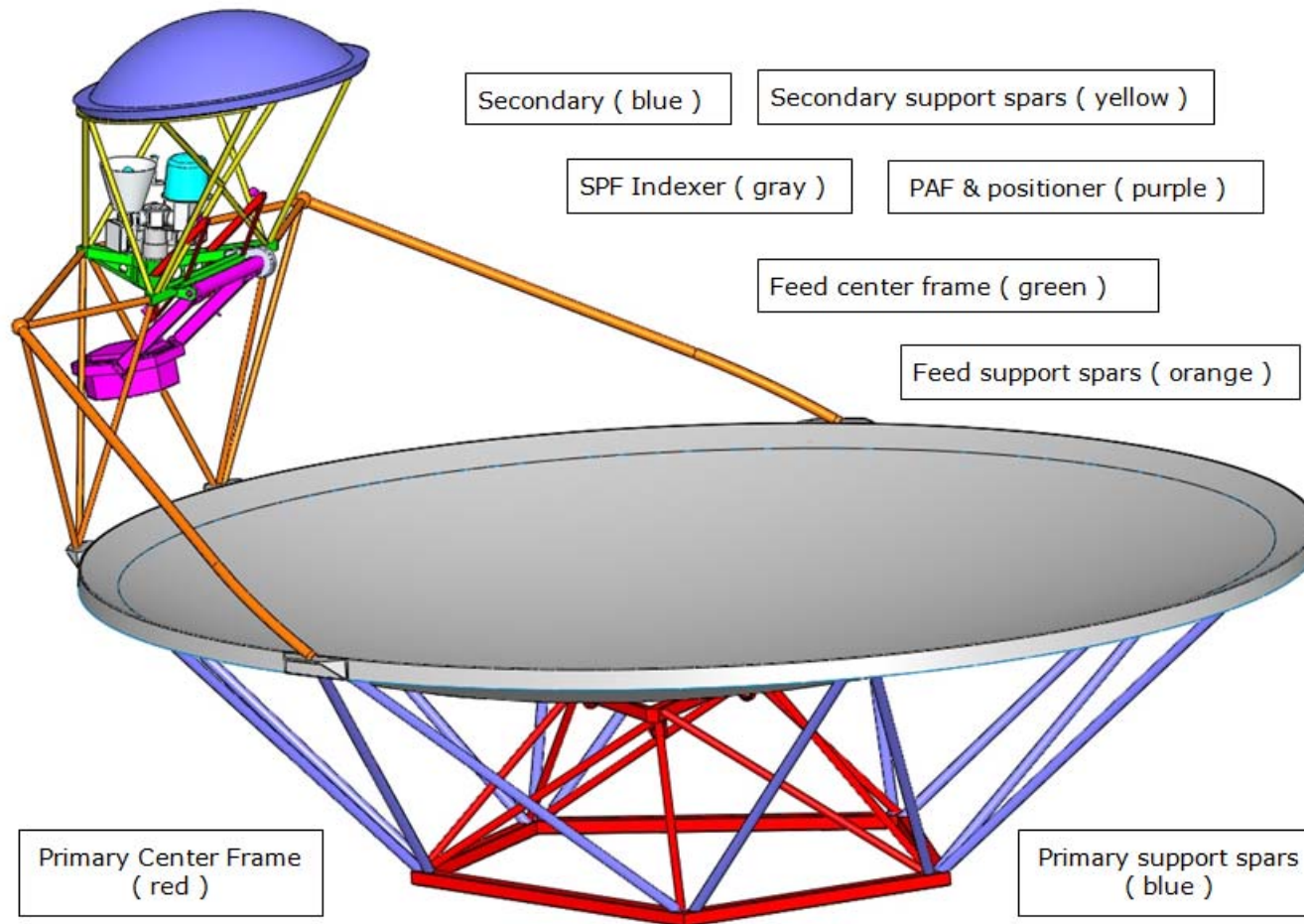


- This shows one concept to accommodate multiple feed payloads on a 'feed indexer', individually replaceable.



Matt Fleming

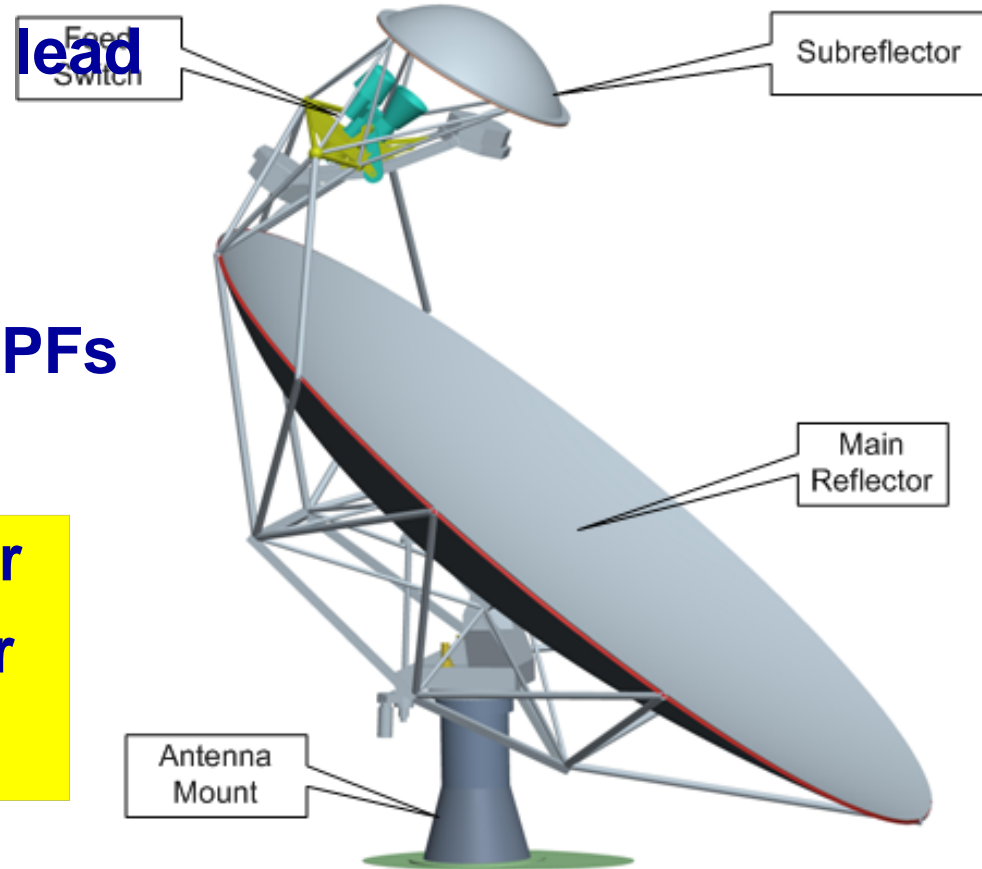
Feed and Support Structure



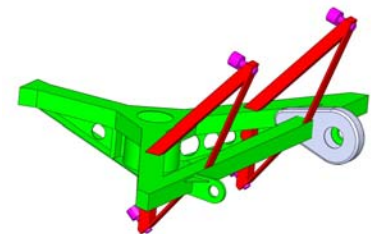
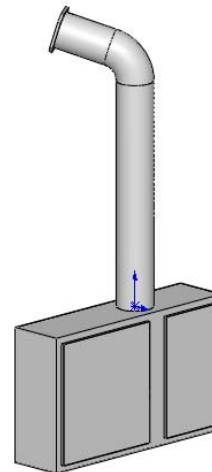
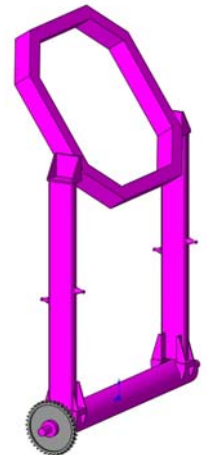
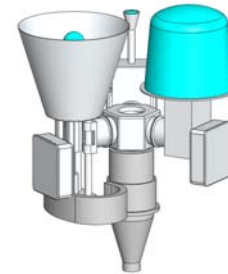
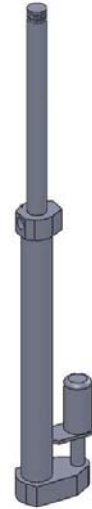
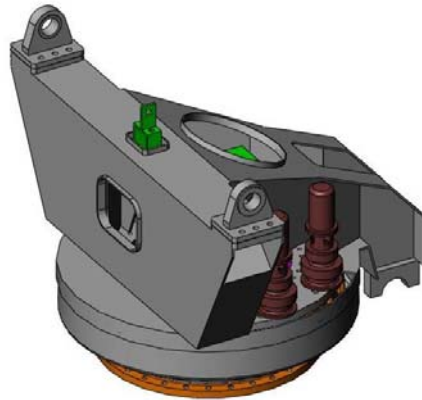
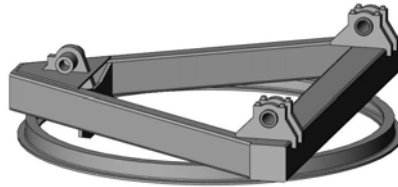
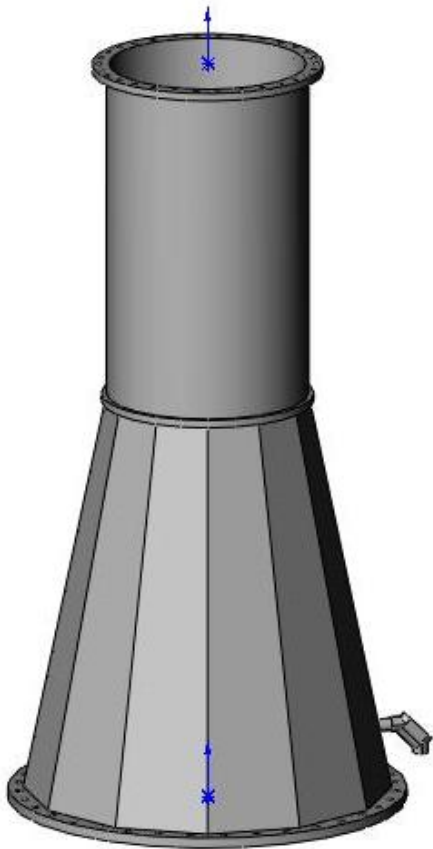
DVAC-1 : Offset design from JLRAT (China)



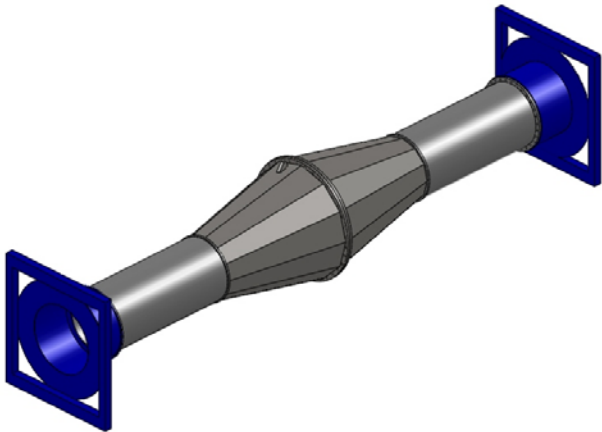
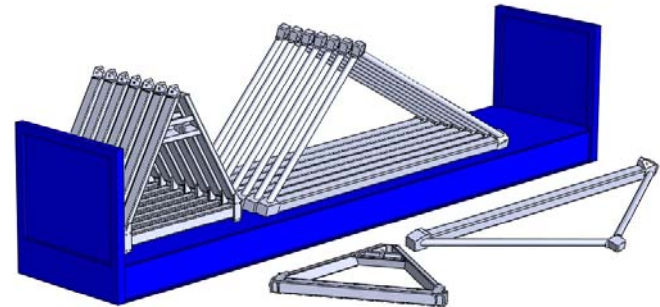
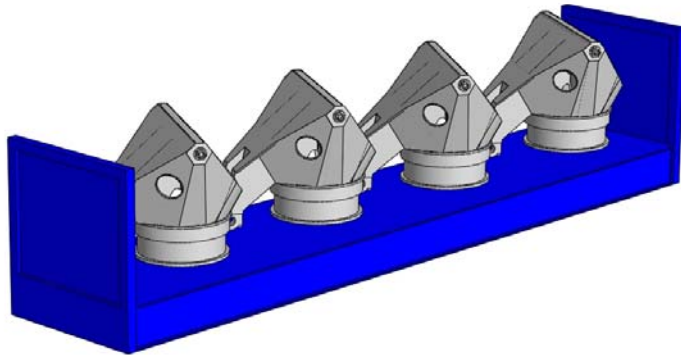
- **Single integrated main reflector**
- **Minimal spar structure**
- **Turning head design with a lead screw elevation actuator**
- **Support and interchange mechanism for a PAF and 3 SPFs or 2 WBFs.**
- **Can be designed with either metal or carbon fibre reflector skins**



Deliverables



Shipments

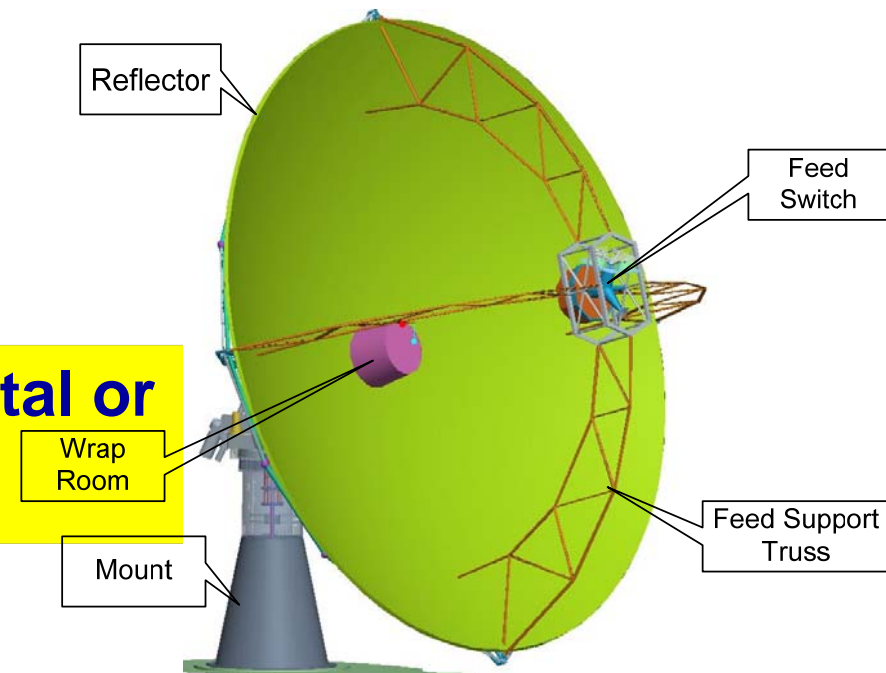


DVAC-2 : Symmetric design from JLRAT (China)



- Single integrated main reflector
- Minimal spar structure
- Turning head design with a lead screw elevation actuator
- Four support legs and interchange mechanism for a PAF and 3 SPFs or 2 WBSPFs.

➤ Can be designed with either metal or carbon fibre reflector skins



4. Concept Design

(2) Structure Design



→ Reflector Design — Main reflector

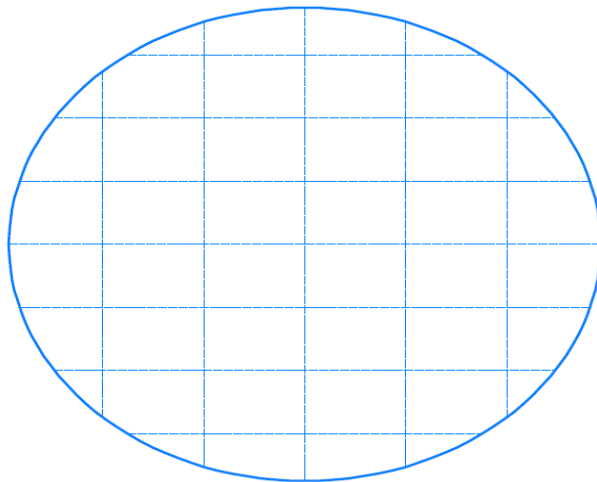
✓ Design 1: Aluminum sandwich structure

Single aluminium panel

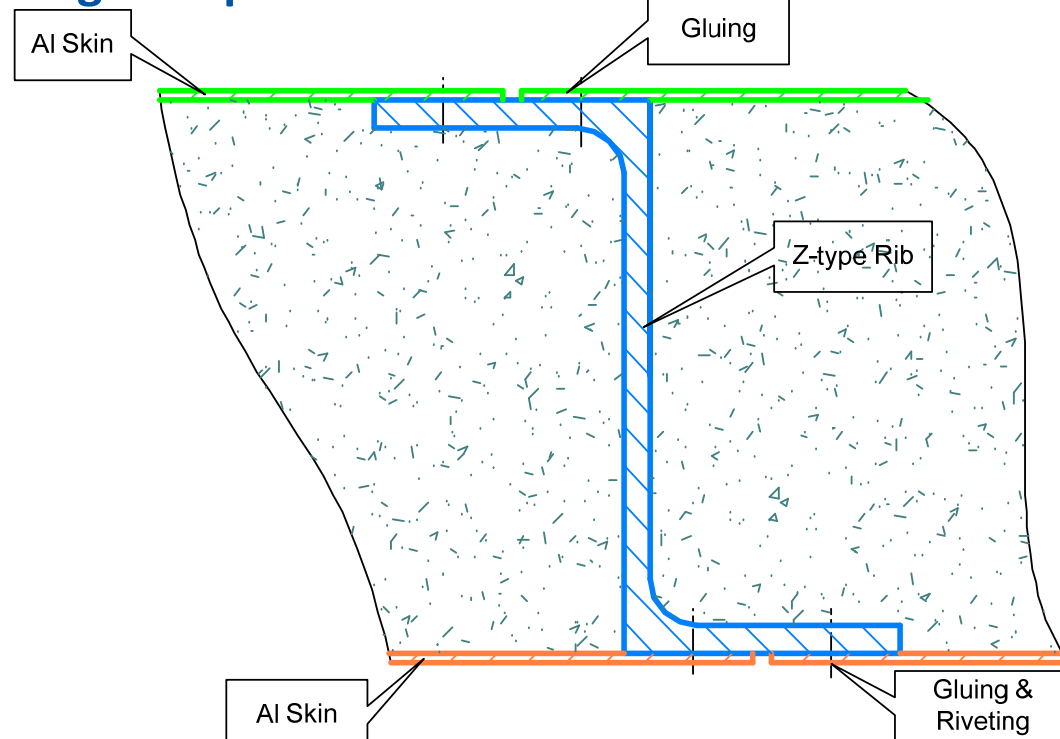
Skins: 2mm (upper)/1mm (lower) in thickness, 2m in width

Skin and ribs are glued through negative pressure method on mould

Surface accuracy $\sigma \leq 0.8\text{mm}$



Rib Configuration



4. Concept Design

(2) Structure Design



→ Reflector Design — Main reflector

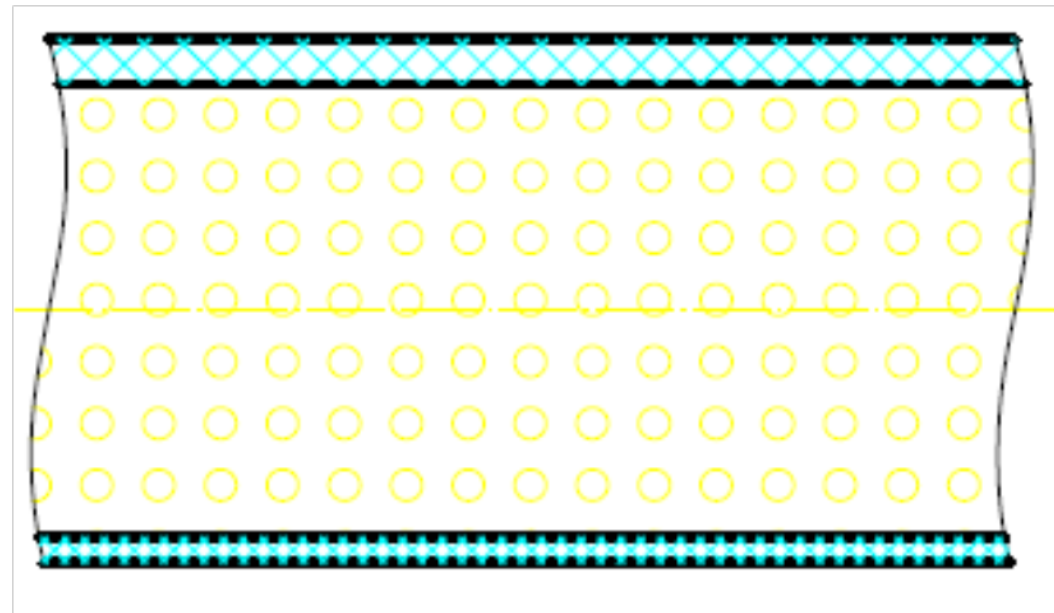
✓ Design 2: Carbon fibre sandwich structure

Single carbon fibre panel

Carbon fibre skins: 1.5mm (top)/1mm (bottom) in thickness

Polyurethane foam: in the middle

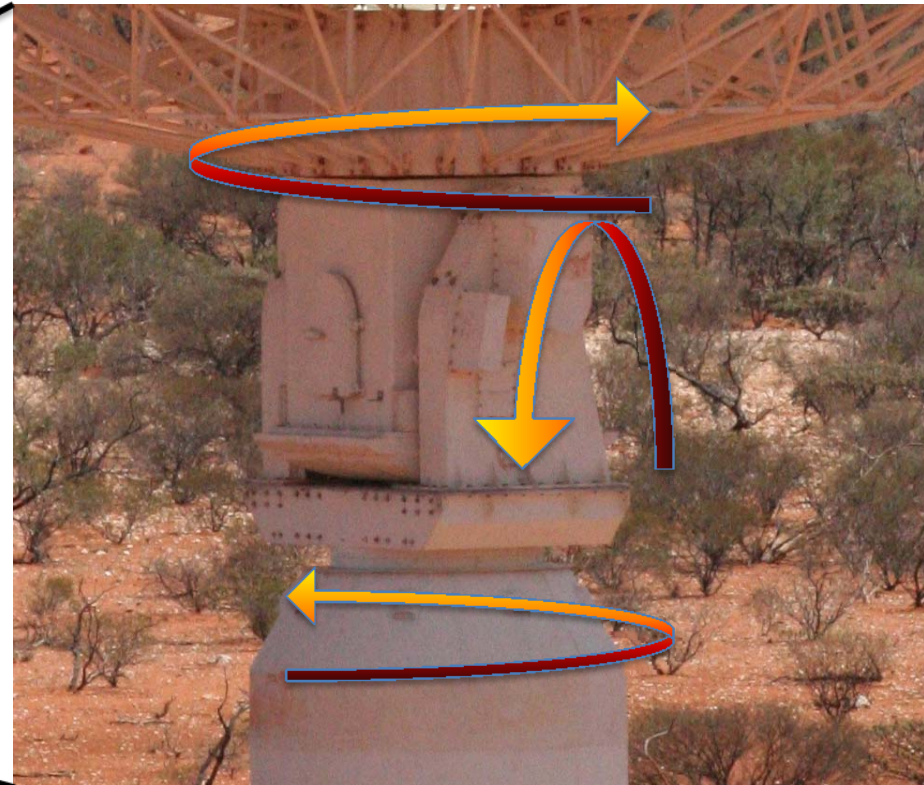
Surface accuracy $\sigma \leq 0.8\text{mm}$



ASKAP 3-axis Antenna Design



3-axis ASKAP antennas in Australia



Exploring the Universe with the world's largest radio telescope

Dish CoDR Panel Members



Roger Norrod (Chair)

National Radio Astronomy Observatory, USA (ret).

Trevor Bird

Antengenuity/CSIRO, Australia.

Peter Dewdney

SKA Program Development Office (SPDO), UK.

Bob Plemel

SED Systems (formerly, ret), Canada.

Tony Willis

National Research Council, DRAO, Canada.

CoDR Panel Recommendations



1. Program in which **imaging experts work with dish designers**
 - modeling and interaction
 - **assess the impact of dish design** on imaging dynamic range.
2. Formal engineering requirements document for the Dish Array,
 - specification document for the dish antenna element.
3. For composite fabrication techniques,
 - **accelerated lifetime testing** in an outdoor environment
 - under conditions similar to the SKA candidate sites.
4. Establish a **clear and uniform methodology to assess all antenna designs.**
5. Develop a test plan for candidate antennas.
 - Build prototypes of the best candidate(s), test, and
 - feed results back into array system models (per Recommendation 1).
6. Establish a **specific set of feeds and receivers to include in SKA1.**
 - Perform accurate cost analysis (total cost of ownership) and develop receiver prototypes.
7. Address the **challenges related to use of cryogenic receiver systems** in the SKA.
8. Trade-off study of **shaped vs. non-shaped (conic section) optics.**

1. Uncertain future availability of cryogenic LNA transistors/MMICs.
2. Lack of cryogenic coolers of adequate reliability and performance that are also operationally affordable.
 - Work on cryogenics has been done for KAT7.
 - Experience with 80K Stirling coolers.
3. Insufficient interaction between imaging and antenna design experts.
4. Current lack of a detailed Operational Plan.

Current Dish Cost Estimates



Cost estimates for 15 m offset Gregorian dishes for the SKA						
quantity basis	600	3000	1 - 250	1 - 250	251 - 3000	251 - 3000
concept	DVA1	DVA1	DVAC1 aluminium	DVAC1 carbon fibre	DVAC1 aluminium	DVAC1 carbon fibre
estimator	DRAO/TDP	DRAO/TDP	NAOC/JLRAT	NAOC/JLRAT	NAOC/JLRAT	NAOC/JLRAT
total cost €	€ 201,557	€ 185,432	€ 198,000	€ 207,000	€ 191,000	€ 199,000
reflector cost €	€ 69,775	€ 64,193	€ 75,000	€ 85,000	€ 73,000	€ 82,000

Cost estimates for 15 m axi-symmetric dishes for the SKA						
quantity basis	1 - 250	1 - 250	251 - 3000	251 - 3000	?	?
concept	DVAC2 aluminium	DVAC2 carbon fibre	DVAC2 aluminium	DVAC2 carbon fibre	Thermoplastic axi-symmetric	Thermoplastic axi-symmetric (future)
estimator	NAOC/JLRAT	NAOC/JLRAT	NAOC/JLRAT	NAOC/JLRAT	ASTRON	ASTRON
total cost €	€ 211,000	€ 220,000	€ 202,000	€ 210,000		
reflector cost €	€ 65,000	€ 75,000	€ 63,000	€ 72,000	€ 106,000	€ 84,800

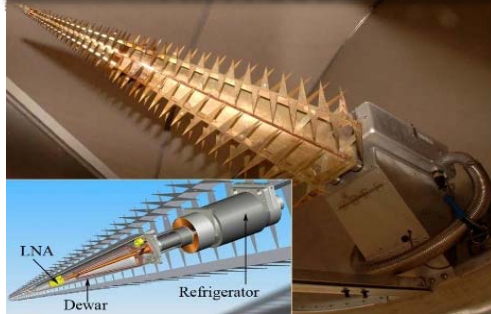
Cost Minimization Principles for Dishes & Payloads



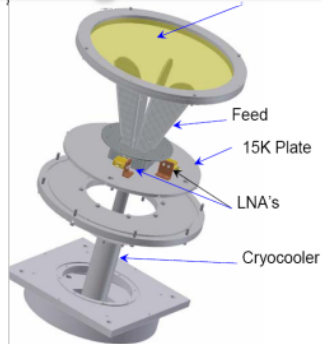
1. Keep the number of dishes as low as possible for a required system sensitivity.
 - Dramatically affects capital and operational system costs.
 - e.g. SKA1 needs only 153 dishes for $A_e/T_{sys} = 6.5 \text{ m}^2/\text{K}$.
 - Dish numbers ripple through the system costs.
2. R&D and spending on noise reduction is very efficient.

Wide Band Single Pixel Feeds

ATA Feed



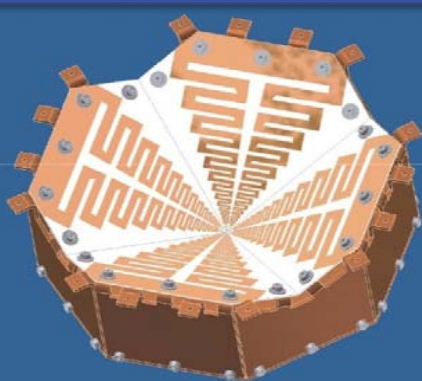
Lindgren Feed



QSC Feed



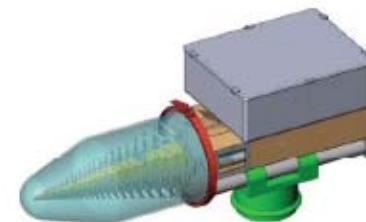
Eleven Feed



Inverted Sinuous Feed



Cooled ATA Feed



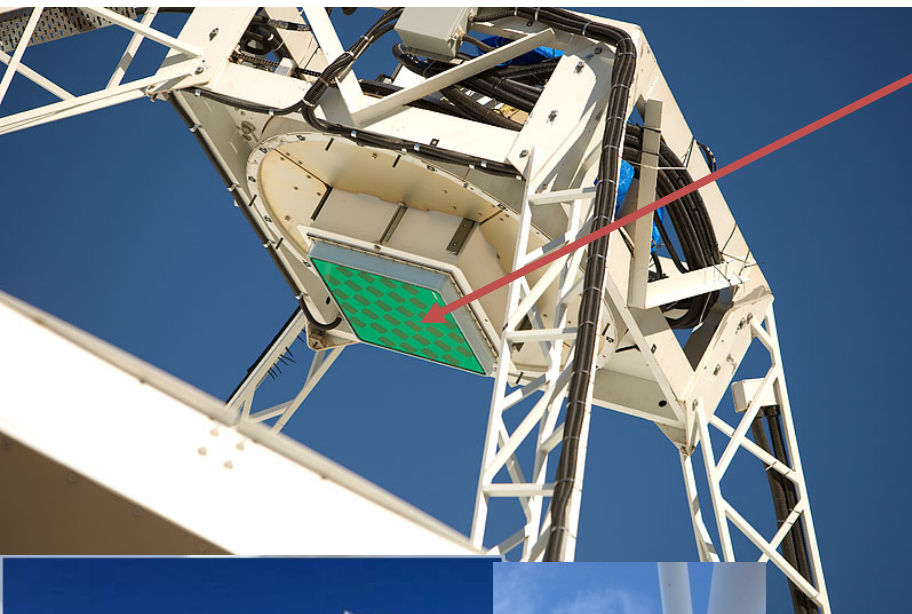
Quad-Ridge Flare Horn



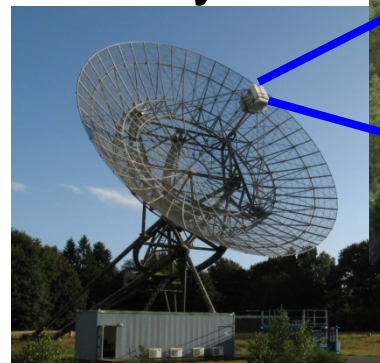
Log-Periodic Log Spiral



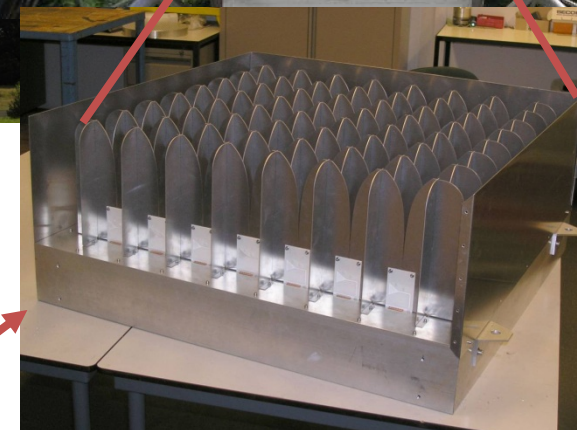
Phased Array Feed Concepts



ASKAP
chequer board
array



APERTIF
(Astron, NL)



Vivaldi arrays



DRAO
Canada

Dipole array
(cooled version)



Risks:



1. 24 slides containing risks presented at Dish Array CoDR
2. Considerable risk:
 - a) Programmatic
 - b) Cost
 - c) Schedule
 - d) Gaps

Risk Summary



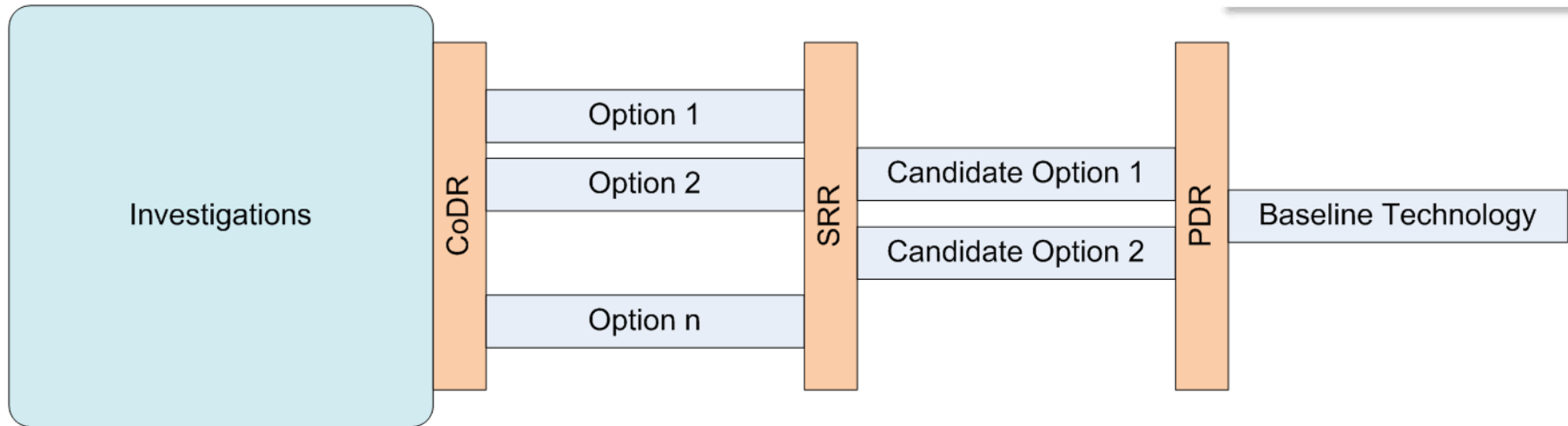
- Dishes
 - Insufficient development funding.
 - Production cost: developing a full understanding.
 - Performance: limiting spatial or spectral dynamic range at full sensitivity.
 - Remote site: maintenance and access.
 - Environment: longevity.
- SPF feed/LNA payloads:
 - Development gaps:
 - not enough attention to development of the best performance feeds
 - Not enough attention to cryogenics: weight, power, maintenance, longevity
 - Power consumption: especially with cryogenics.
 - Cost of ownership: maintenance costs (esp cryogenics).
 - EMC, RFI

Risk Summary (cont'd)



- PAF Sub-System
 - PAF System Performance
 - Weight and Volume
 - Power Consumption
 - EMC and RFI Compliance
 - Development Timeline
 - Costs (full system costs)
- SPF Receiver
 - Analog Performance
 - Gain/Phase stability
 - Analog bandshape stability
 - ADC
 - Technology Maturity
 - Power Consumption
 - EMC, RFI.

Narrowing of Options



- System Engineering Approach => happens at every level of the system hierarchy.
 - see SKA System Eng. Management Plan and other docs.

CoDR = Concept Design Review

SRR = System Requirements Review

PDR = Preliminary Design Review

End